

## Environnemental challenges in Aeronautics

« Research in Advanced Aeronautics  
Combustion Chambers »

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## Engine contribution to environmental objectives

### Objectives ACARE 2020

(Advisory Council for Aeronautics Research in Europe)



- 50% reduction of noise
- 80% reduction of NOx emission
- 50% reduction of CO2



ATM Contribution

Aircraft Contribution

Engine Contribution

- 6 db Reduction of noise
- 60 to 80% Reduction of Nox
- 20% Reduction of kerosene consumption

## Physical principles for an environmental compromise

### Increase the pressure ratio

- Reduction of fuel / CO<sub>2</sub> ratio by an enhancement of the energetic efficiency
- Reduction of unburnt HC and CO emissions
- Increase of NO<sub>x</sub> emissions due to an increase of the combustion chamber temperature
- Increase of the maintenance cost

### Increase of the by-pass ratio of the engine

- Reduction of fuel / CO<sub>2</sub> ratio by an enhancement of propulsive efficiency
- Reduction of noise by reduction of the rotation speed of the fan, the pressure ratio, exhaust velocity of gases
- Increase of weight, shear and maintenance cost

### Enhancement of aerodynamics and weight reduction

- Efficiency level already reached
- Reduction of CO<sub>2</sub> by enhancing the fuel efficiency and the thrust
- Reduction of noise by reducing the needed thrust
- Reduction of No<sub>x</sub> emissions by reducing the combustion temperature

## Research challenges

### Development of new physical models for:

- Development of chemical kinetic models (combustion, NO<sub>x</sub> and particles production ,..)
- Development of numerical tools (mesh, equation solving,..) for chamber design

### Need of experimental studies for numerical validation

- Validation of physical models in academic experimental configurations
- Development of adapted diagnostic techniques, using often laser (hostile mediums, weak or any optical acces for measurements (velocity, turbulence, temperature, chemical composition, pressure, noise, location of reactive zones, .. )
- *Full scale tests*

### Developpement of new alloys

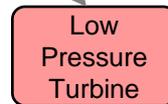
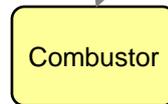
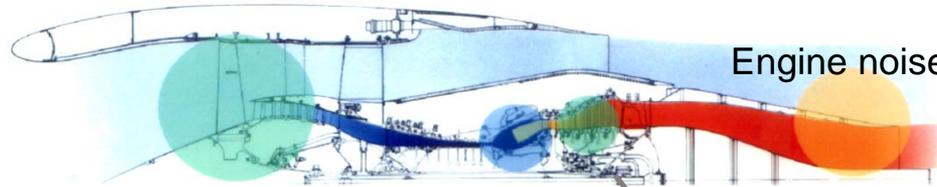
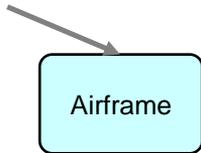
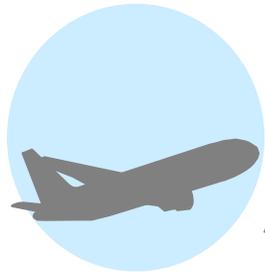
- Weight reduction
- Material behaviour during thermal cycles

## Identified Research challenges

1. Chemical kinetics
2. NO<sub>x</sub> formation:  
New Concepts of combustion chambers  
LPP regime Lean premixed prevaporized (Air exces)  
RQL Combustion Rich Quenching Lean
3. Turbulent combustion
4. Chamber design – numerical tool development
5. Wall cooling
6. Fuel atomisation
7. Gas and particle radiation
8. Alternative kerosene
9. Hydrogen or natural gas
10. <sup>turbines</sup> New combustion modes in aeronautical turbines

## Conclusion : research challenges

- Optimisation of the 2-phase combustion chambers for a reduction of **CO<sub>2</sub>, NO<sub>x</sub>, CO and particle emission**
- Optimisation of the **wall cooling** for chambers and turbine **blades**
- Reduction of **noise**



Documents SNECMA

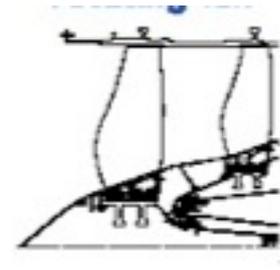
Nozzle with chevrons



Enhancement of entrance



Contra-rotative fan



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- Optimisation of the **wall cooling** for chambers and turbine **blades**
- Reduction of **noise**
- Development of **new alloys** (weight reduction, material behavior at high temperature, thermal cycles,...)

Composite fan case



Highly loaded compressor



Light weight Composite mixer



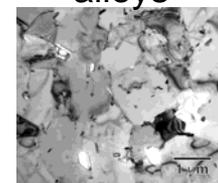
Next-gen RTM composite fan



Ceramic-matrix composite blade



Novel titanium alloys



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